

Solutions Manual for Elasticity in Engineering Mechanics, Third Edition

Arthur P. Boresi and Ken P. Chong, and James D. Lee



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2-4.6

$$\begin{aligned}
 I_2(\bar{C}) &= \bar{C}_{ij} \bar{C}_{ji} \\
 &= a_{i\alpha} a_{j\beta} a_{j\gamma} a_{i\kappa} C_{\alpha\beta} C_{\gamma\kappa} \\
 &= a_{\kappa\alpha} a_{\gamma\beta} C_{\alpha\beta} C_{\gamma\kappa} \\
 &= C_{\alpha\beta} C_{\alpha\beta} = I_2(C)
 \end{aligned}$$

$$\begin{aligned}
 I_3(\bar{C}) &= \bar{C}_{ij} \bar{C}_{jk} \bar{C}_{ki} \\
 &= a_{i\alpha} a_{j\beta} a_{j\gamma} a_{k\kappa} a_{k\chi} a_{i\eta} C_{\alpha\beta} C_{\gamma\kappa} C_{\chi\eta} \\
 &= a_{\alpha\eta} a_{\gamma\beta} a_{\chi\kappa} C_{\alpha\beta} C_{\gamma\kappa} C_{\chi\eta} \\
 &= C_{\alpha\beta} C_{\beta\chi} C_{\chi\alpha} = I_3(C)
 \end{aligned}$$

2-4.7

By expanding the matrix and finding the determinant, it is straightforward to prove that

$$I_A = A_{11} + A_{22} + A_{33} = A_{kk}$$

$$\begin{aligned}
 II_A &= \begin{vmatrix} A_{22} & A_{23} \\ A_{32} & A_{33} \end{vmatrix} + \begin{vmatrix} A_{11} & A_{13} \\ A_{31} & A_{33} \end{vmatrix} + \begin{vmatrix} A_{11} & A_{12} \\ A_{21} & A_{22} \end{vmatrix} \\
 &= \frac{1}{2} \left\{ (A_{kk})^2 - (A_{ij} A_{ji}) \right\}
 \end{aligned}$$

$$\begin{aligned}
 III_A &= \begin{vmatrix} A_{11} & A_{12} & A_{13} \\ A_{21} & A_{22} & A_{23} \\ A_{31} & A_{32} & A_{33} \end{vmatrix} \\
 &= \frac{1}{6} \left\{ 2(A_{ij} A_{jk} A_{ki}) - 3(A_{kk} A_{ij} A_{ji}) + (A_{kk})^3 \right\}
 \end{aligned}$$

2-15.7 proceeding as in problem (2-15.6), we obtain

$$u = A\left(\frac{x^2}{2} - z^2x\right) - A\frac{y^2}{2} + C_3y + C_2z + C_4$$

$$v = A\left(xy - \frac{y^2}{2}\right) - A\frac{z^2}{2} - C_3x + C_5z + C_6$$

$$w = A\left(yz + \frac{z^2}{2}\right) + A\frac{x^2}{2} - C_2x - C_5y + C_7$$

For $u = v = w = 0$ at $x = y = z = 0$, $C_4 = C_6 = C_7 = 0$

For $\bar{w} = 0$ at $x = y = z = 0$, $C_2 = C_3 = C_5 = 0$

$$\therefore u = A\left(\frac{x^2}{2} - \frac{y^2}{2} - xz\right)$$

$$v = A\left(xy - \frac{y^2}{2} - \frac{z^2}{2}\right)$$

$$w = A\left(\frac{x^2}{2} + yz + \frac{z^2}{2}\right)$$

2-15.8 $u_{\alpha} = C_{\alpha\beta} x_{\beta} = C_{\alpha\gamma} x_{\gamma}$

$$E_{\alpha\beta} = \frac{1}{2} [u_{\alpha,\beta} + u_{\beta,\alpha} + u_{\theta,\alpha} u_{\theta,\beta}]$$

$$\therefore 2E_{\alpha\beta} = C_{\alpha\beta} + C_{\beta\alpha} + C_{\theta\alpha} C_{\theta\beta}$$

(a) If we take $C_{\alpha\beta} = -C_{\beta\alpha}$, then

$E_{\alpha\beta} = \frac{1}{2} C_{\theta\alpha} C_{\theta\beta}$ is quadratic in $C_{\alpha\beta}$. Hence the quantities $E_{\alpha\beta}$ vanish.

(b) In case (a) $E_{\alpha\beta}$ is quadratic in $C_{\alpha\beta}$. Hence, the quantities $E_{\alpha\beta}$ vanish. If we discard quadratic terms in $E_{\alpha\beta}$, then we discard everything.

2-15.9

Given: $u = -(1 - \cos \varphi)x - y \sin \varphi$

$$v = x \sin \varphi - (1 - \cos \varphi)y$$

$$w = 0; \quad \varphi = \text{const}$$

Then $e_{11} = \frac{\partial u}{\partial x} = -(1 - \cos \varphi)$ $e_{22} = \frac{\partial v}{\partial y} = -(1 - \cos \varphi)$

$$e_{33} = \frac{\partial w}{\partial z} = 0$$

$$e_{12} = \frac{1}{2} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) = 0$$

$$e_{13} = \frac{1}{2} \left(\frac{\partial w}{\partial x} + \frac{\partial u}{\partial z} \right) = 0, \quad e_{23} = \frac{1}{2} \left(\frac{\partial w}{\partial y} + \frac{\partial v}{\partial z} \right) = 0$$

cont'd

3-6.2

In terms of $\sigma_1, \sigma_2, \sigma_3$ (principal stresses)

$$S_x = \sigma_x - \sigma_m = \sigma_1 - \sigma_m = S_1,$$

$$S_y = \sigma_y - \sigma_m = \sigma_2 - \sigma_m = S_2$$

$$S_z = \sigma_z - \sigma_m = \sigma_3 - \sigma_m = S_3$$

$$\sigma_m = \frac{1}{3}(\sigma_1 + \sigma_2 + \sigma_3)$$

Hence,

$$T_d = \begin{pmatrix} S_1 & 0 & 0 \\ 0 & S_2 & 0 \\ 0 & 0 & S_3 \end{pmatrix}$$

$$I_{2d} = S_1 S_2 + S_1 S_3 + S_2 S_3 = -\frac{1}{6}[(S_1 - S_2)^2 + (S_2 - S_3)^2 + (S_3 - S_1)^2]$$

Since $S_1 + S_2 + S_3 = 0$

$$I_{3d} = S_1 S_2 S_3$$

3-6.3

By Eq (3-3.14) the stress normal to a plane

with direction cosine n_α is $\sigma_{nn} = \sigma_{\alpha\beta} n_\alpha n_\beta$. For

$(n_1^2, n_2^2, n_3^2) = (l^2, m^2, n^2) = (\frac{1}{3}, \frac{1}{3}, \frac{1}{3})$ relative to

principal axes $\sigma_{nn} = \sigma_{oct}$. Also, then $\sigma_{11} = \sigma_1, \sigma_{22} = \sigma_2$

$\sigma_{33} = \sigma_3$, and $\sigma_{12} = \sigma_{23} = \sigma_{31} = 0$

$$\therefore \sigma_{oct} = \sigma_1 n_1^2 + \sigma_2 n_2^2 + \sigma_3 n_3^2 = \frac{\sigma_1 + \sigma_2 + \sigma_3}{3} = \frac{\sigma_x + \sigma_y + \sigma_z}{3} = \sigma_m$$

Since, $\sigma_x + \sigma_y + \sigma_z = \sigma_1 + \sigma_2 + \sigma_3$ is invariant.

4-7.5

(a) simple tension:

$$\text{By Eqs. (4-7.1), } J_1 = \epsilon_x + \epsilon_y + \epsilon_z = \frac{\sigma}{3\lambda + 2G} = (1 - 2\nu) \frac{\sigma}{E}$$

(b) pure shear:

$$\text{since } \epsilon_x = \epsilon_y = \epsilon_z = 0, J_1 = 0$$

Problem Set 4-8

The two problems in this set are quite lengthy, but instructive. The solutions are not presented here, but may be obtained from the authors upon request.

7-10.5 cont'd

term to Eq.(4), V in the form

$$V = \frac{G\theta}{2} \left[\frac{r^2 \cos 2\beta}{\cos \alpha} + a^2 \sum_n A_n \left(\frac{r}{a}\right)^\lambda \cos \lambda \beta \right] \quad (5)$$

The first additional term in the bracket of Eq.(5) reduces to $\frac{G\theta r^2}{2}$ for $\beta = \pm \frac{\alpha}{2}$. Therefore, the Σ term in Eq.(5) must vanish for $\beta = \pm \frac{\alpha}{2}$.

Hence, we require

$$\lambda = \frac{n\pi}{\alpha} \quad \text{for } n = 1, 3, 5, \dots \quad (6)$$

Hence,

$$\phi = \frac{G\theta}{2} \left[-r^2 \left(1 - \frac{\cos 2\beta}{\cos \alpha} \right) + a^2 \sum_{n=1,3,5,\dots}^{\infty} A_n \left(\frac{r}{a}\right)^{\frac{n\pi}{\alpha}} \cos \frac{n\pi\beta}{\alpha} \right] \quad (7)$$

To satisfy Eq.(3), Eq.(7) yields

$$\sum A_n \cos \frac{n\pi\beta}{\alpha} = 1 - \frac{\cos 2\beta}{\cos \alpha} \quad (8)$$

By the Method of Fourier series (section 7-10), we find

$$A_n = \frac{16\alpha^2}{\pi^3} (-1)^{\frac{n+1}{2}} \frac{1}{n(n + \frac{2\alpha}{\pi})(n - \frac{2\alpha}{\pi})} \quad (9)$$

Hence, the result given in Problem 7-10.5 is obtained.